

# An Integrated Risk Management Framework

-Introducing the Triple-Triplets Concept for SMA

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# Why An Integrated S&MA Management Framework Is Important?

- A systematic approach to resolving S&MA issues as identified in the CAIB report:
  - > "Risk information and data from hazard analysis are not communicated effectively to the risk assessment and mission assurance process ..."
  - > "System safety engineering and management is separated from mainstream engineering ...."
  - > "Over the last two decades, little to no progress has been made toward attaining integrated, independent, and detailed analysis of risk ...."
  - ➤ No process addresses the need to update hazard analysis when anomalies occur."
  - ➤ Need of "a disciplined, systematic approach to identifying, analyzing, and controlling hazards ..."
- NPG 7120.5A, enacted in April, 1998, requires that "
  The program or project manager shall apply risk
  management principles ....



- The complexity of NASA' new challenges in CEV/CLV design development and its successful operation necessitates an integrated S&MA management process
- Hazard, Safety, Reliability and Risk are integral elements to comprehensive SMA management of any complex engineered systems.
- Need of an integrated process for combining hazard analysis with PRA, along with other system safety & reliability techniques for Systematic SMA Management.
- Utilization of a systems engineering thought process SMA function itself within a space program/project is a closed loop adaptive control system.



# Why An Integrated Risk Management Framework is Important for S&MA? (Cont'd)

- Space Exploration Beyond LEO Has Brought New Reality & Tough Challenges for NASA
  - > Fundamentally new
  - Greater Complexity
  - Multifaceted
  - > Public Scrutiny
  - > Uncertainty

# Level/Scope of Integrated Risk Management

• What is Integrated Risk Management?

Integrated risk management is the integration of the management of risk at each level of management into all business and strategic planning and decision-making processes.

- > Technological risk aspect
- Programmatic risk aspects
- > Enterprise / Organizational risk aspects
- > The total risk management



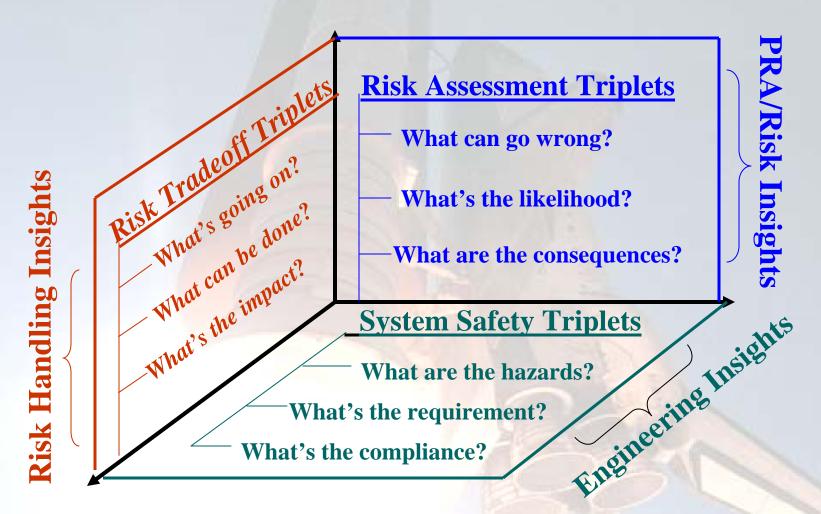
# Comprehensive & Total Risk Management



## A Triple-Triplets (Double-T) Conceptual Framework

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- A Systems Engineering based Process for SMA



## Why the Triple-Triplets Concept is Needed?

- A set of fundamental concept in assurance engineering
- A pack of guiding principles in risk management
- A system engineering-based SMA process in a nutshell
- A consolidated framework combines all method/techniques
- An easy to understand/communicate questions for us all
- An integrated tool handles both technical/programmatic risks



# Why the Triple-Triplets Concept is Needed? (Cont'd)

Conceptual Differences of System Hazard, Risk, Safety, Reliability:

- <u>HAZARD</u> System threat existed that can cause potential damage & harm. A necessary condition for risk but not absolute condition for risk or damages.
- <u>RISK</u> An integrated measurement of consequence of a undesired event occurrence. Not necessarily a mathematically measurable quantity
  - Technical risk vs Programmatic risk;
  - Risk vs Problem
- <u>SAFETY</u> Assurance or level of confidence in accident/damage prevention & control. The system safety concept is the application of systems engineering and mgmt to the process of hazard, safety & risk analysis to identify, assess & control associated hazards while designing or modifying systems, products, or services.
- RELIABILITY Assurances of expected proper functioning of equipment, systems, hardware or software component as well as human performances etc. Low reliability must induce high risk but low risk not necessarily come from high reliability.

# The Paradox of Safety, Reliability & Risk Taking

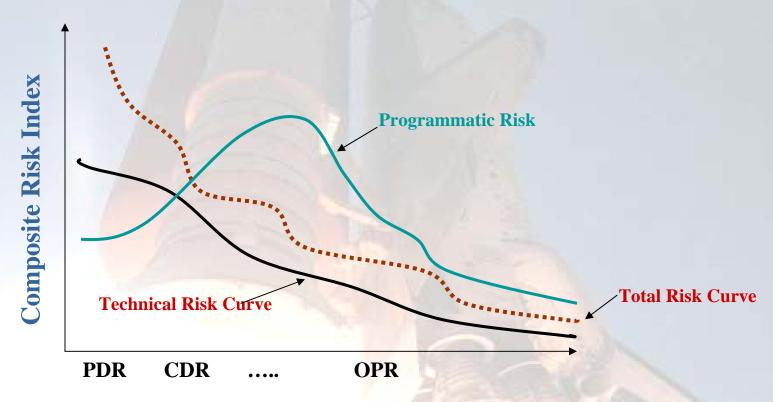
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Program/Project managers need to be very clear:

- High reliability, high redundancy and high cost design & space operations do not necessarily mean high safety and greater mission successes
- "It's how you manage it stupid!"
  - How to identify, analyze
  - How to make risk trade-off decisions with multi-objectives (often conflicting objectives)
  - How to focus & allocate resources
  - How to track, communicate & handle risks
- Major Challenge exist on how to best trade-off, consolidate (or aggregate) and handling all types of risks:
  - Technical & Programmatic risks;
  - Political, Social, Environmental & Organizational risks;
  - Cost & Schedule & Safety & Mission Assurance risks



# **Illustration of Synthesized Risk Curves**



Program/Project Life Cycle (Time)





# The System Safety Triplets

### - A Safety Engineering Process

#### 1. What are the hazards?

**Failure source identifications** (hardware/software/human/organization/external)

Hazard analysis/Hazard ranking using risk index matrix (semi-quantitative FTA)

FMEA/FMECA and CILs on root cause identification & initiator ranking

### 2. What are the requirements?

Develop safety requirements & goal - when & where to impose?

What are the organizational hierarchy & assurance for hazard control?

Process for ensuring reliability, maintainability, supportability & inspections

## 3. What's the compliances?

<u>Safety audit & regulatory mechanisms</u> for compliance & verificationsProcess for documentation control and hazard/risk communicationsCulture for two-dimensional (vertical/horizontal) Risk/Hazard communications



# The Risk Assessment Triplets

### - A PRA Process To Gain Risk Insights

## 1. What can go wrong?

Risk identification (for all credible & significant hazards)

**Hazards & Initiating event identification** 

Scenario development, enumeration and structuring

#### 2. What's the likelihood?

Risk quantification & measurement

Reliability & Data assessment

Risk evaluation & uncertainty assessment

**Risk ranking & importance measures** 

### 3. What are the consequences?

Risk mitigation & Damage assessment

Failure & success criteria evaluations



# The Risk Trade-off Triplets

#### - A Risk-Informed Decision Process

## 1. What's going on?

Trend Analysis RM & Risk-based performance monitoring/evaluation Indicator technology - quantitative/qualitative trend/time series assessment) Accident Sequence Precursor (ASP) identification & evaluations Data mining & statistical anomalies/near-miss assessment Communication of issues & problems

#### 2. What can be done?

<u>Trade-off studies</u> using insights from both PRA & Hazard Analysis (HA) What options are available & what are their associated trade-offs? Multi-objective, optimized cost-benefit analysis (CBA) & decision making

### 3. What's the impact?

**Impact assessment** of current mgmt decisions on future options (risk reduction)

Impact of risk control evaluations of risk mgmt activities on safety improvement

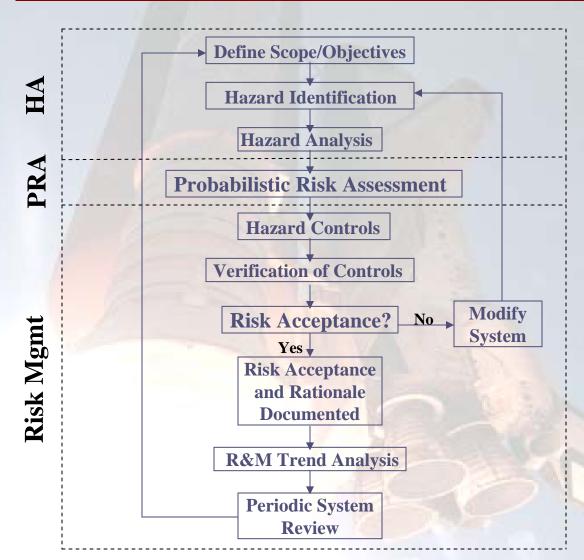
# The Double-T Concept

- A Simple Prescription for Mission Success:
  - In Risk management, there is no crystal ball, no fortune teller, but there are guiding principles:
  - If the fundamental 9 key questions (as represented by the Triple-triplet concept) are asked at least once a day
  - If asked frequently at every level of program hierarchy and project milestones by managers, design engineers, SMA engineers, operational technicians and everyone in the process
  - Then the chances are: everyone' life in our risky space business will be much easier, healthier and happier than ever before



## The "Double-T" S&MA Management Concept

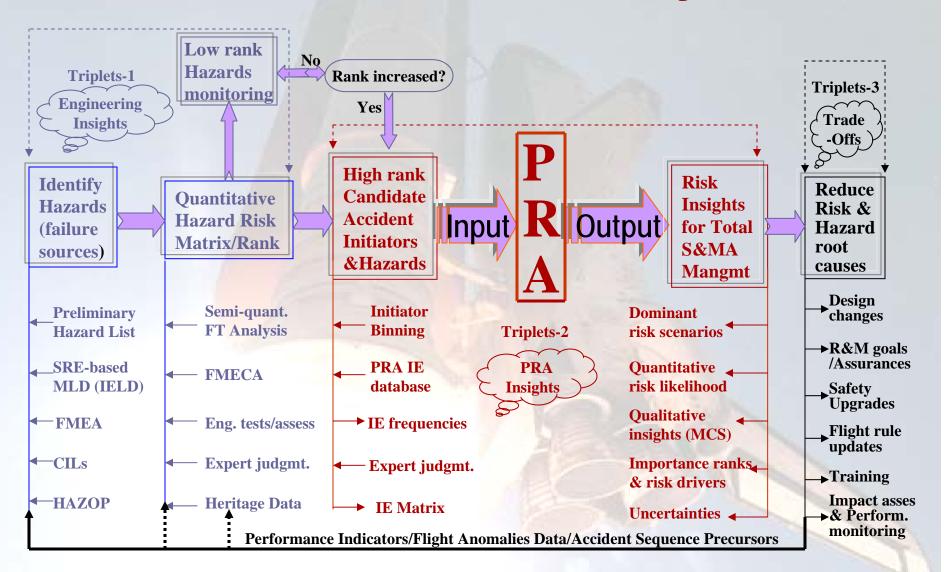
**A Simplified Example Systems Engineering Process** 



# The "Double-T" S&MA Management Framework

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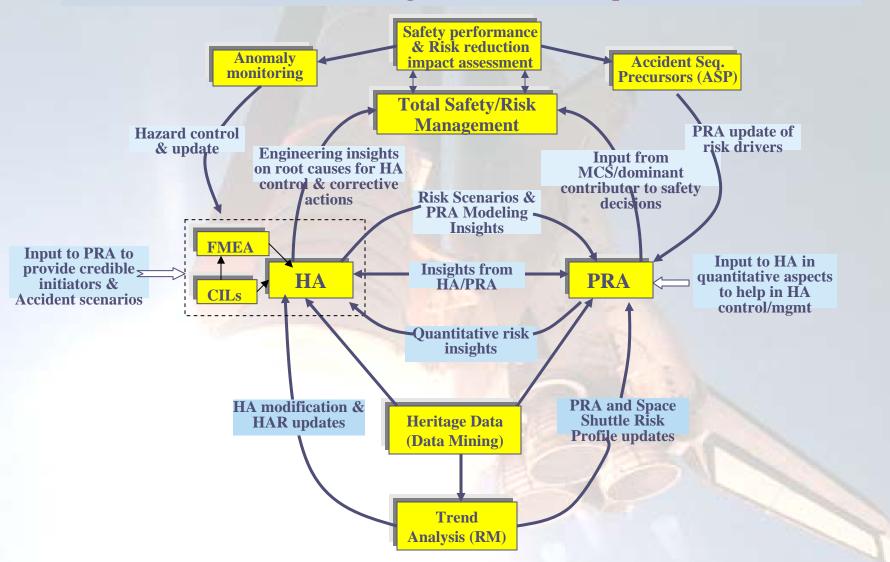
- Role of HA & PRA in the "Double-T" S&MA Mgmt Process



# The "Double-T" S&MA Management Framework (Cont

- An Integrated Process for Combining Hazard Analysis with PRA for Safety and Risk Management (The SMA Spider)

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## The "Double-T" S&MA Management Framework – Key Elements

# A Systematic & Comprehensive Approach for Hazard Identification/Analysis

A systematic accident initiator identification using SRE (Scenariostructured Risk Envelope) concept

A method to combine & incorporate Hazard Analysis (HA) process into PRA

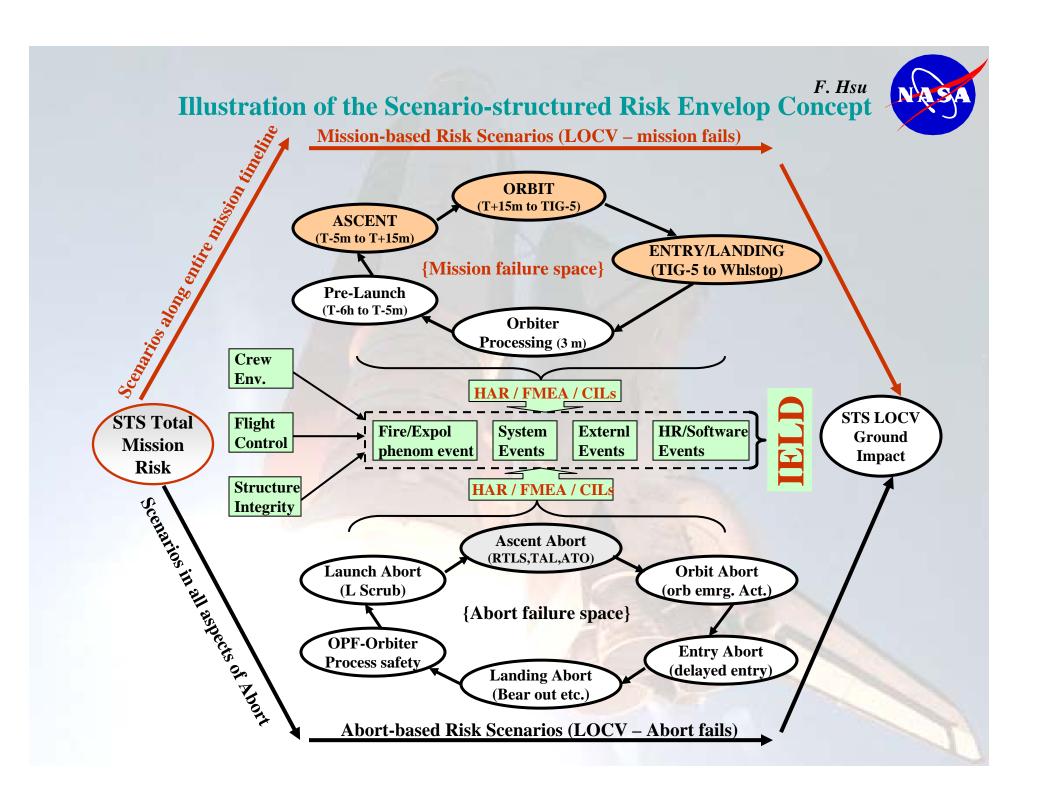
A Systematic HA Approach which ensures completeness in searching, analyzing, ranking and reporting of hazard/failure sources for S&MA

A improved HA process, which becomes a key element of the proposed total Risk-informed S&MA management framework based on "Double T" concept



# The "Double-T" S&MA Management Framework – Key Element (Cont'd)

- ♣ The Scenario-structured Risk Envelop (SRE) Concept for Searching & Identifying Hazards
  - The SRE adhere to the concept of "enveloping the risk" in completeness
  - The philosophy behind the SRE concept finding accident before accident find us!
  - SRE the need for completeness in PRA (all LOCV potentials are considered)
  - A systemic approach for searching candidate initiating events. searching the entire spectrum of all dimensions of failure space along phases, functions, and mission timeline



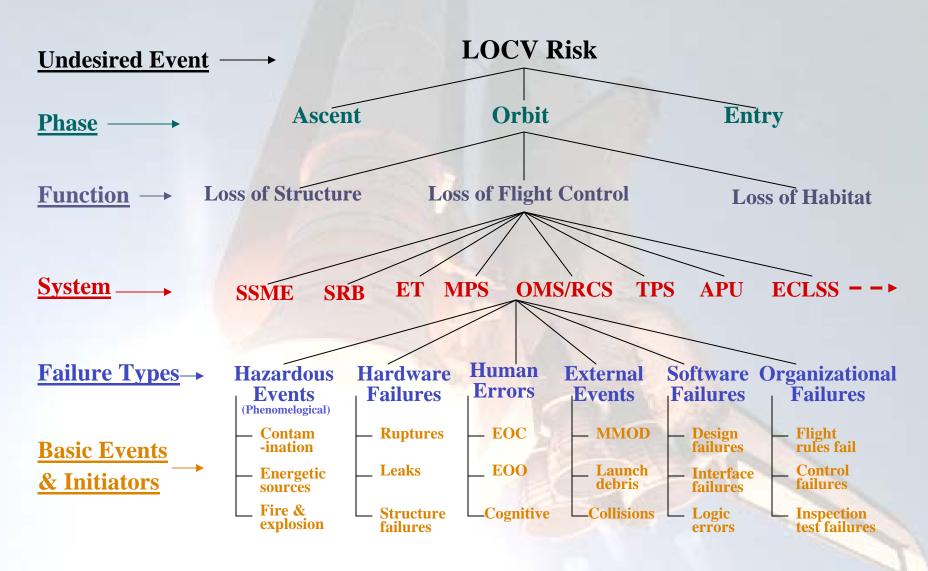


# The "Double-T" S&MA Management Framework – Key Element (Cont'd)

- The SRE-based Initiating Event Logic Diagram (IELD)
  - IELD a matrix formed Initiating Event Logic Diagram. An effective tool for managing, documenting and representing vast amount of candidate hazardous initiating events for risk model considerations
  - A computerized IELD database format can be conveniently established
  - Similar to conventional MLD Top down, summary logic diagram. It identifies and categorizes a more complete set of IEs.
  - SRE concept incorporates a functional thought process and provides a bridge to relate NASA's vast engineering assessment databank (HARs/FMEA/CILs)

## An Example Hierarchy of SRE-based Initiating Event Log Diagram (IELD) for Systematic Hazard Identification

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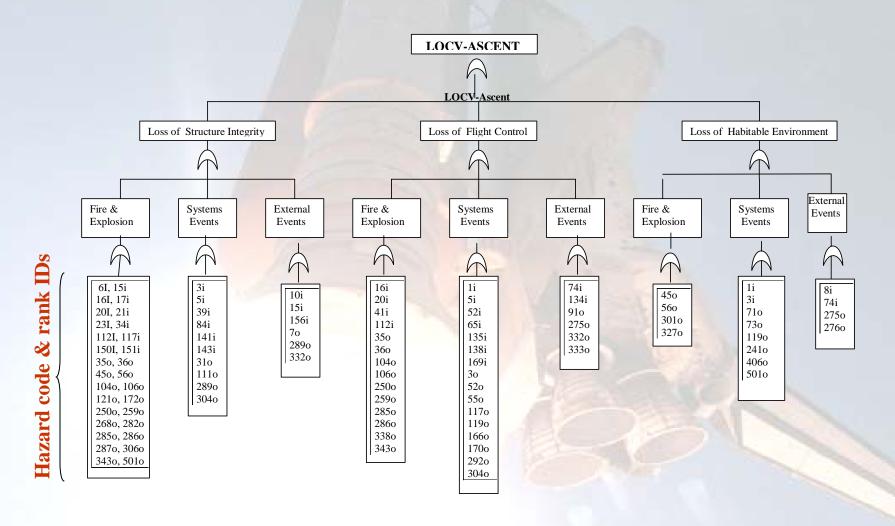
# **An Example Matrix-based Representation of IELD**

		The M	Iatrix Represo	entation of M	lodularized N	ALD Sub-tree	s for the Integ	rated Shuttle	PRA ML	D }	
	Top -Level Func failures	Loss of	Structure Integrity	A	Lo	ss of Flight Control	A	Loss of Habitable Environment			
	Mission Phases	Fire/Explosion	Systems Events	External Events	Fire/Explosion	Systems Events	External Events	Fire/Explosion	Systems Events	External Events	
	LOCV-PreLch (LOCV During PreLaunch)	LOCV-PreLch-LS-FirExp	LOCV-PreLeth-LS-SysEvi	LOCV-PreLch-LS-ExtEvt	LOCV-PreLeb-FC-FirExp	LOCV-PreLch-FC-SysEvt	LOCV-PreLeh-FC-ExiEvt	LOCV-Prelch-EN-FirExp	LOCV-PreLeh-EN-SysEvt	LOCY-PreLeb-EN-ExtEvt	
I Phases A	LOCV-Ascent (LOCV During Ascent)			.OCV-Ascent-LS-SysEvt LOCV-Ascent-LS-ExtEvt		LOCV-Ascent-FC-SysEvt	LOCV-Ascent-FC-ExtExt	LOCV-Ascent-EN-FirExp	LOCV-Ascent-EN-SysEv1	LOCV-Ascent-EN-ExtExt	
Mission-Based	LOCV-Orbit (LOCV During Orbit)	LOCV-Orbit-LS-FirExp	LOCV-Orbit-LS-SysEvt	LOCV-Orbit-LS-ExtEvt LOCV-Orbit-FC-FirExp		LOCY-Orbit-FC-SysEvt	LOCV-Orbit-FC-ExtExt	LOCV-Orbit-EN-FirExp	LOCV-Orbit-EN-SysEvt	LOCV-Orbit-EN-ExtEvt	
2	LOCV-DesLnd (LOCV During Des/Land)	LOCV-DesLnd-LS-FirExp	LOCV-DesLnd-LS-SysEvt	LOCV-DesLnd-LS-ExtEvt	LOCV-DesLnd-FC-FirExp	LOCV-DesLnd-FC-SysEvt	LOCV-DesLnd-FC-ExtEvt	LOCV-DesLnd-EN-FirExp	LOCV-DesLnd-EN-SysEvt	LOCV-DesLnd-EN-ExtEvt	
nases A	LOCV-AbrtAsnt (LOCV During Asnt Abort)	LOCY-AbrtAsnt-LS-FirExp	LOCV-AbritAsnt-LS-SysEvt	LOCV-AbrtAsmi-LS-ExtExt	LOCV-AbrtAsmt-FC-FirExp	LOCV-AbriAsni-FC-SysEvi	LOCV-AbrtAsm-FC-ExtEvt	LOCV-AbrtAsni-EN-FirExp	LOCY-AbrtAsni-EN-SysEvi	LOCV-AbrtAint-EN-ExtEvt	
Abort-Based Phases	LOCV-AbrtOrbt (LOCV During Orbit Abort)	LOCV-AbrtOrbi-LS-FirExp	V-AbrtOrbi-LS-FirExp LOCV-AbrtOrbi-LS-SysEvt		LOCV-AbrtOrbt-FC-FirExp	LOCV-AbriOrbi-FC-SysEvi	LOCV-AbrtOrbt-FC-ExtEvt	LOCV-AbriOrbi-EN-FirExp	LOCV-AbriOrbi-EN-SysEvi	LOCV-AbriOrbi-EN-ExiEvi	
Αl	TOCV-AbrtDeLd (LOCV During Descent & Landing Abort)	LOCV-AbriDeLd-LS-FirExp	LOCV-AbriDeLd-LS-SysEvi	LOCV-AbriDeLd-LS-ExtExt	LOCV-AbriDeLd-FC-FirExp	LOCV-AbriDeLd-FC-SysEvi	LOCV-AbriDeLd-FC-ExtEvi	LOCV-AbriDeLd-EN-FirExp	LOCV-AbriDeLd-EN-SysEvi	LOCV-Abri DeLd-EN-ExtEst	

## **A Graphical Representation of IELD**

#### A Graphical Representation of A Partial Initiating Logic Diagram (IELD)

(For ASCENT Phase of the Integrated Shuttle PRA)



## List of Accident Initiating Events Identified in the IELD

(MPS Related Example Initiators)

USA MLD Missi			Threatene					Drob		Doforon	A m a live t		Individual Hazard Departation				
USA Hazard	M L D in itia				<u> </u>			Hazard		_	<u>rob</u>	Referen		alyst	Individual Hazard Description		
		Phas	System	-6	-6	edi	<u>a F</u>	unc	tion	Category		<u>Category</u>	egory	ce ESD	Remarks		
Number	t	e	yste/	OUS	ns			F/P	/P Type	Sev	Like	<u>Names</u>	FT/E Justi				
	٠		ارد	PRACORSEGUE					.,,,					fic a ti			
				88.										o n			
															Ignition of Flammable Atmosphere at the ET / Orbiter LH2 Umbilical		
INTG 006	4	PΑ	MPS	LOCV	SΙ			Р	FE	Α	С		FT		Disconnect Assembly		
															Isolation of the ET from the Orbiter MPS or SSMEs (17 inch valve bursts		
	6		MPS				ΗE		FE	Α	С				open under pressure from ET)		
INTG 016			MPS		SI			Р	FΕ	Α	С		FT		Ignition Sources Igniting Flammable Fluids in the Aft Compartment		
INTG 019			MPS	LOCV		FC		F	SE	Α	С			ΜE	Premature shutdown of one or more SSME's		
INTG 020	18	Α	MPS	LOCV	SI	FC		Р	FE	Α	С		FT		Hydrogen Accumulation in the Aft Compartment During Ascent		
															Contamination in the Integrated Main Propulsion System (which clogs		
	20		MPS		SI			Р	FE	Α	С		FT		the system)		
INTG 034			MPS		SI			Р		Α	С			nbk	Autoignition in High Pressure Oxygen Environment (in MPS)		
	392		MPS	LOCV	L.	FC	Ш	F	FE	Α	С		FT		Loss of MPS/SSME He supply pressure		
INTG 042			MPS		SI			Р	-	Α	С		FT		Turbopump Fragmentation During Engine Operation		
INTG 112			MPS		SI			Р	FE	Α	С		FT		H2/O2 Component Leakage During Ascent/Entry		
INTG 112			MPS	LOCV	SI			Р	FE	Α	С		FT		H2/O2 Component Leakage During Ascent/Entry		
INTG 168	81	PΑ	MPS	LOCV	SI	FC			ΕE	Α	С		FT		Flammable Atmosphere in the ET Intertank (see 238)		
					l			_		١.				1	Hydrogen Accumulation in the Orbiter Compartments During RTLS/TAL		
ORBI035	102	A D	MPS	LOCV	SI	FC		Р	FE	Α	С			Abt	Abort		
					L.		l I	_		١.					Ignition of Orbiter Fluids Entrapped in the TCS Materials (aft		
ORBI045	107	PAOD	MPS	LOCV	SI	FC	ΗE	Р	FE	Α	С		FT		compartment)		
0.001.400					١			_	0 =	١.					Overpressurization of the Orbiter Aft Fuselage Caused by the Failure of		
ORBI 108	133	PAOD	MPS	LOCV	SI			Р	SE	Α	С		FT	<u> </u>	an MPS Helium Regulator or Relief Valve  Loss of Structural Integrity Due to Overpressurization of the Mid and/or		
0.001.000					١			Р	0 =	١.					, ,		
ORBI 278	187	PAOD	MPS	LOCV	SI			Р	SE	Α	С		FT	<u> </u>	Aft Fuselage Fire/Explosion in the Orbiter Aft Compartment Caused by MPS		
0.000	005	D 4		1001				Р									
ORBI 306 ORBI 338			M P S M P S		SI SI			P P	F E F E	A	С		FT		Propellant Leakage / Component Rupture GO2 External Tank Pressurization Line as MPS/APU Ignition Source		
ORBI 338	219	РА	M P S	LOCV	51	FC	$\vdash$	Р	FE	Α	С		FT	-	Fire/Explosion in the Orbiter Aft Compartment Caused by Contamination		
0.001.040	004	РА	MDC	1001	SI			Р	FE	_	_						
ORBI343 INTG 085			MPS	LOCV	SI	FC		P	FE	A A	c d		FT	-	in the Main Propulsion System Feed System Ignition of Flammable Atmosphere at T-0 Umbilicals		
IN I G U65	44	Р	WFS	LUCV	31		Н	Р	ГС	А	u		ГІ		Malfunction of the LH2 and LO2 T-0 Umbilical Carrier Plate Resulting in		
INTG 089	15	PΑ	MPS	LOCV	SI			F	SE	Α	d		FT	1	Damage to Shuttle Vehicle		
	71		MPS	LOCV	SI			•		A	d		r .	Abt	Potential Geysering in the LO2 Feed Line (Tsat = boiling point)		
INTG 153			MPS			FC		г Р	SE	A	d		-		Premature Separation of Orbiter T-0 Umbilical Carrier Plate		
111 10 100	, 5		IVI 1' 3	LOCV	01	0	$\vdash$		J L	_	u u	1	-	17.01	Overpressurization of LO2 Orbiter Bleed System or LH2 Recirculation		
INTG 167	8.0	Р	мрѕ	LOCV	SI	FC		Р	SE	Α	d			Abt	System		
ME-FG3P		l ·	MPS		SI	<del>                                     </del>		г Р	SE	A	d	1	FT	7.51	geysering of LOX (MPS) (see 71)		
ME-FG6S			MPS		SI			г Р	SE	A	d	1	l '	Abt	abnormal thrust loads		
ME-FG8M			MPS		SI			P	SE	A	d	1	FT	7,51	thrust oscillations leading to pogo (see 3)		
ORBI 248		PAOD		LOCV	SI	FC		P	FE	A	d		FT	<del>                                     </del>	Fire/Explosion in GOX Pressurization System		
ME-FA1S			MPS			FC	Н		F E	C	c		<del></del>	<b>-</b>	hydrogen fire/explosion external to aft compartment (see 21)		
2 17.10	0.10				٠.					٦	~						
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					-		$\vdash$										
					1						1	<u> </u>		1			

# Example Accident Initiator Bins (Hazard Categories) Developed from I

(There can be a logic mapping between PRA model elements and each of the Hazard categories identified

	Phenomnelogical Initiating Event	Hazard# Identified in IMLD
3in-1:	Eiro/ovalosion from ovternal logkage/runture	
31n-1:	Fire/explosion from external leakage/rupture  Ignition at ET/Orb Umbilical	INTG 006
	Ignition Sources in Aft Compt*	INTG 000
	Hydrogen Accumulation in Aft**	INTG 010
	Ingnition at T-0 Umbilical	INTG 085
	H2/O2 Leakage during Ascent	INTG 112
	H2/O2 Leakage at ET Intertank	INTG 168
	External H2 Leakage	ME FA1S
	H2 in Aft during RTLS/TAL	ORBI 035
	H2/O2 in Aft**	ORBI 306
	GO2 Press Line as Ignition Source*	ORBI 338
3in-2:	Contamination of LH2/LO2 Systems	<u> </u>
JIII-Z.	Contamination of LH2/LO2 Systems	INTG 023
	Fire/Explosion due to Contam. in LH2/LO2 Systems	ORBI 343
	The Explosion due to Contain. In E112/EC2 Gystems	OKBI 343
3in-3:	System Overpressurization	
	Overpress of LO2 Bleed/LH2 Recirc System	INTG 167
	ET Overpressurization	P.01
	MPS H2/O2 manifold overpressure	???
	MPS propellant line overpressrization	INTG167
3in-4:	Aft Overpressurization	
	Aft-overpress due to 750 Reg/850 RV	ORBI 108
	Generic Mid/Aft Compartment Overpressurization	ORBI 278
Bin-5:	GO2 Autoignition	
	GO2 Autoignition	INTG 034
	Ignition of fluids caught in TCS	ORBI 045
	GO2 Autoignition	ORBI 248
Bin-6:	LO2 Water-Hammer	
	GO2 Geyser during Loading/Detank	INTG 153
	GO2 Geyser during Loading/Detank	ME FG3P, A
	Functional Initiating Event	Hazard# Identified in IMLD
3in-7:	Structural Failure of Umbilicals	
	Isolation of ET from Orb/SSME/Ground	INTG 009
	Physical Malfunction of T-0 Umbilical	INTG 089
	ET GH2/GO2 pressure not maintained	ORBI338, S.05
	ET Separation Failure (premature Sep. & ORB ET recontact)	ORBI289, INTG051, P.07
	MPS O2 prevalve fails to close at MECO	INTG039
3in-8:	Loss of SSME NPSP	
	Loss of LO2 NPSP @ MECO	INTG 039
	MPS failure to maintain propellant supply to SSME	???
3in-9:	Loss of GHe	
	Loss of GHe Supply Press	INTG 041/ORBI108
	Loss of GHe for SSME Intermediate Seal Purge	?
<b>.</b>	LO2 Pogo	
Bin-10:	SSME Pogo	ME FG8M



# The "Double-T" S&MA Management Framework – Key Elements (Cont'd)

## **Proposed Hazard Analysis Worksheet Format**

Hazard T Hazard_N			1		rol_Status: ard risk index:	/			
Element: System: Subsystem	n:	Phase:				Date: Analy Doc.#	st: F. H		
Hazard & Control #	Hazard Descriptio n	Cause factors	Potential Effects	Hazard risk index	PRA Coverage (IE/BE/Model)	Control Recom' d	Effect of Recm'd	Verific a-tion of control	Status of control
INTG37		AB					300		
		C				X		1/	

# The "Double-T" S&MA Management Framework – Key Elements (Cont'd)



#### Proposed Hazard Risk Assessment Matrix & Semi-quantitative Risk Index

Hazard Title& Hazard/Control No. INTG 037 # Causes: A,B,C,D,E,F Total Hazard Risk Index: 2.1E-5 Severity: high

	Category ncy Bins	Consequence Severity Index - Based on worst case (LOCV) conditional likelihood)								
`*	nission) or each bin)	Negligible 1 (.0001)	Minimal <b>2</b> (.001)	Marginal 3 (0.01)	Critical 4 (0.1)	Catast 5 (1.0)				
1E-2 ~1E00 50th: 1E-1	5 Likely > 1E-2	1E-5 (1/100000)	1E-4 (1/10000)	1E-3 (1/1000)	1E-2 (1/100)	1E-1 (1/10)				
1E-4 ~ 1E-2 50th: 1E-3	4 Probable 1E-4~1E-2	1E-7	1E-6	1E-5	1E-4 A*B*C	1E-3 (1/1000)				
1E-6 ~ 1E-4 50th: 1E-5	3 Infrequent 1E-6 ~ 1E-4	1E-9	1E-8 E+F	1E-7	1E-6	1E-5 (1/100000)				
1E-8 ~ 1E-6 50th: 1E-7	2 Unlikely 1E-8 ~ 1E-6	1E-11	1E-10	1E-9	1E-8 A+C+G	1E-7				
1E-10~1E-8 50 <sup>th</sup> : 1E-9	1 Remote 1E-10 ~ 1E-8	1E-13	1E-12	1E-11	1E-10	1E-9				

 $HIV = \Sigma M_{i,j} \text{ where } M_{i,j} = \{\Sigma X_k \text{ if } X_k \text{ is additive; } \Pi X_k \text{ if } X_k \text{ is multiplicative} \} \text{ is } HIV \text{ in cell } \{i,j\}$ 

# The "Double-T" S&MA Management Framework – Key Elements (Cont'd)

(Examples To be Provided)

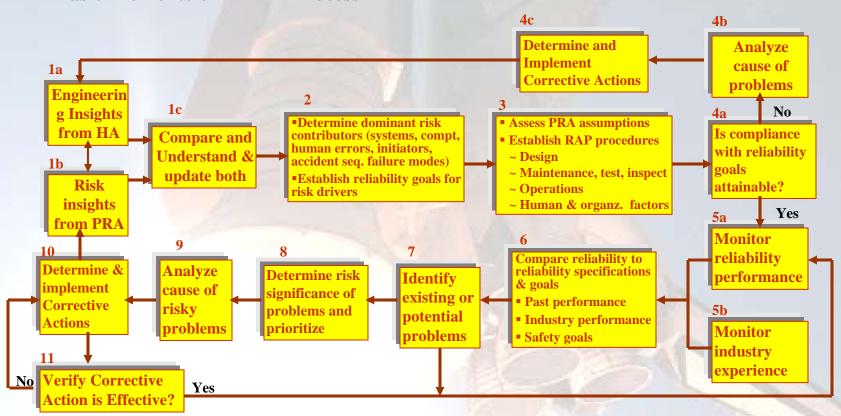
- **→** Hazard Identification Based on innovative SRE Concept
- ➤ Innovative Hazard Analysis Use of Semi-quantitative Risk Matrix
- > Hazard Ranking Methodology
- > Relationship, Mapping & Control of Hazard in PRA
- **▶** Use of Accident Sequence Precursor (ASP) Analysis technique
- > Utilization of a RAP (Reliability Assurance Program) process



# The "Double-T" S&MA Management Framework – Key Elements (Cont'd)

- A Proposed Reliability Assurance (RAP) Program

Basic Elements of A RAP Process





## **Concluding Remarks**

- A systematic Triple-triplet concept has been introduced based on the systems theory to facilitate an integrated risk management framework for SMA
- Key to integrated risk management is the system-based thought process in risk identification, assessment and decision-making. It's not necessarily depending on the format of the physical process itself
- Effective integrated risk management plan and implementation must imbed within every phases of a program/project activities along its entire life cycle
- Adequate use of PRA and analytical decision-making methodology can play a vital role in successful integrated risk management
- A systematic hazard identification based on the SRE technique along with the proposed semi-quantitative risk matrix can be a more effective risk management approach over the conventional risk matrix method